

Fig. 2 Idealized collision process in rarefied gas field;  $U \ll \bar{c}$ ,  $\mu_0 = M_0$

tion  $\psi(x)$  consequently is determined by the solution of the "wave" equation:

$$\nabla^2 \psi + (8\pi^2 M_0 E_0 / \beta_0^2) \psi = 0 \quad (15)$$

The free body satisfies a corresponding momentum equation of the form

$$\frac{1}{2\pi i} \frac{\partial \psi}{\partial x} \pm \frac{2\omega_0}{U} \psi = 0 \quad \text{or} \quad \frac{1}{2\pi i} \frac{\partial \psi}{\partial x} \pm \frac{M_0 U}{\beta_0} \psi = 0 \quad (16)$$

If the free body possesses a potential energy  $V$ , the total energy may be written as  $E_T = \frac{1}{2} M_0 U^2 + V$ . Hence,  $M_0 U = (1/\alpha \bar{c})(E_T - V)$ , and Eq. (16) can be written in the alternate form

$$\frac{\beta_0 \bar{c}}{2\pi i} \frac{\partial \psi}{\partial x} \pm \frac{1}{\alpha} (E_T - V) \psi = 0 \quad (17)$$

Equation (15) is a one-dimensional classical analog of the Schrodinger equation, which also may be written in the form

$$\nabla^2 \psi + (8\pi^2 M_0 / \beta_0^2) (E_T - V) \psi = 0 \quad (18)$$

if the particle possesses a potential energy. Equation (17) is a classical analog of the free particle momentum equation and, if  $V$  is identified with an energy  $M_0 \bar{c}^2$ , is a classical analog of the Dirac relativistic wave equation for a free particle.<sup>8,9</sup>

Further details of the characteristics of the idealized monatomic rarefied gas field described by approximate solutions of the Boltzmann equation will be described in forthcoming papers. The observations that transverse disturbances can propagate in such fields,<sup>3,5</sup> that the vector forms of the equations are very similar to the Maxwell equations,<sup>6</sup> and that classical analogs of the equations of quantum mechanics may exist suggest that the significance of these linearized rarefied gas field approximate solutions is much greater than previously has been appreciated.

### References

- Grad, H., "On the kinetic theory of rarefied gases," *Commun. Pure Appl. Math.* 2, 331-407 (December 1949).
- Ai, D. K., "Small perturbations in the unsteady flow of a rarefied gas based on Grad's thirteen moment approximation," *Guggenheim Aeronaut. Lab., Calif. Inst. Tech. Rept.* 59 (September 20, 1960).
- Yang, H. and Lees, L., "Rayleigh's problem at low Mach number according to the kinetic theory of gases," *J. Math. Phys.* 35, 195-235 (October 1956).
- Logan, J. G., "Propagation of thermal disturbances in rarefied gas flows," *AIAA J.* 1, 699-700 (1963).
- Logan, J. G., "A further note on propagation of transverse disturbances in rarefied-gas flows," *AIAA J.* 1, 943-945 (1963).
- Logan, J. G., "Rarefied-gas field equations for plane shear disturbance propagation," *AIAA J.* 1, 1173-1175 (1963).

<sup>7</sup> White, H. E., *Introduction to Atomic Spectra* (McGraw-Hill Book Co. Inc., New York, 1934), pp. 14-56.

<sup>8</sup> Schiff, L. I., *Quantum Mechanics* (McGraw-Hill Book Co. Inc., New York, 1949), pp. 30, 311-327.

<sup>9</sup> Lindsay, R. B. and Margenau, H., *Foundations of Physics* (John Wiley and Sons, Inc., New York, 1949), p. 505.

## Invariant Two-Body Velocity Components and the Hodograph

RUDOLF PEŠEK\*

Czechoslovak Academy of Sciences,  
Prague, Czechoslovakia

### Nomenclature

- $e$  = eccentricity of conic figure
- $F$  = attracting focus of conic figure
- $h$  = specific angular momentum
- $r$  = radial distance between gravitational centers of orbital and celestial bodies
- $V$  = orbital velocity
- $V_h$  = horizontal invariant velocity component
- $V_p$  = invariant velocity component normal to the lines of apsides,  $V = V_h + V_p$
- $V_r$  = radial velocity component
- $V_\theta$  = horizontal velocity component,  $V = V_r + V_\theta$
- $V_x$  = velocity component parallel to the apsidal line
- $V_y$  = velocity component normal to the apsidal line,  $V = V_x + V_y$
- $\beta$  = flight path angle
- $\theta$  = true anomaly
- $\mu$  = gravitational parameter of the two-body system

IN a note published recently,<sup>1</sup> Cronin and Schwartz draw the readers' attention to the "little used property of two-body orbital motion, that the velocity vector at any point can be resolved into two components of invariant magnitude." They write also that "no inferences seem to have been made that there is any practical application which takes advantage of the invariance of these two velocity components."

In a series of articles presented recently,<sup>2,4,5</sup> Altman and Pistiner analyzed different two-body problems by use of the special hodograph. At the 2nd International Symposium on Rockets and Astronautics in Tokyo in 1960, Fang-Toh-Sun introduced also a special hodograph for the solution of orbital problems.<sup>3,6</sup>

The purpose of this note is to show the application of the invariance of two velocity components to the derivation of both the classical (Hamilton) and the special hodograph.

Consider a two-body conic trajectory with focus  $F$ . The velocity  $V$  can be resolved (Fig. 1): 1) into two components parallel and normal to the line of apsides,  $V = V_x + V_y$ ; 2) into two components along and normal to the radius vector from the focus,  $V = V_r + V_\theta$ ; and 3) into two components of invariant magnitude  $V_h$  normal to the radius vector and  $V_p$  normal to the line of apsides,  $V = V_h + V_p$ .

Here, according to Ref. 1,  $V_h = \mu/h$ , a constant and everywhere normal to the radius vector, and  $V_p = (\mu/h)e$ , a constant and everywhere normal to the line of apsides.

From the triangle of velocities  $\triangle ABS$ , one has the equation of classical (Hamilton) hodograph:

$$V_x^2 + (V_y - V_p)^2 = V_h^2$$

In the plane  $V_x, V_y$  the hodograph is a circle; its center is placed on the  $V_y$  axis at a distance  $V_p$  from the origin, and its radius is equal to  $V_h$  (Fig. 2).

Received March 20, 1963.

\* Chairman, Commission on Astronautics. Member AIAA.

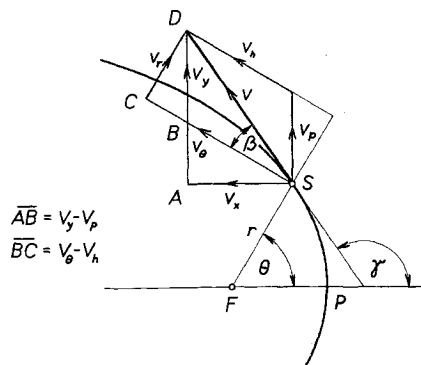


Fig. 1 Conic trajectory and velocity components

From the triangle of velocities  $\triangle BCD$ , one has the equation of special hodograph:

$$(V_\theta - V_h)^2 + V_r^2 = V_p^2$$

In the plane  $V_\theta, V_r$ , this hodograph is a circle; its center is placed on  $V_\theta$  axis at a distance  $V_h$  from the origin, and its radius is equal to  $V_p$  (Fig. 3).

It is clear that both the classical and special polar hodographs are valid for all classes of conic figures, i.e., elliptic, parabolic, and hyperbolic. Different classes of conic figures are provided by attendant changes of  $V_p$  and  $V_h$ .

For elliptical orbit,

$$e < 1 \quad V_p < V_h$$

for parabolic orbit,

$$e = 1 \quad V_p = V_h$$

and for hyperbolic orbit,

$$e > 1 \quad V_p > V_h$$

#### Conclusions

The orbital velocity in the two-body orbital motion can be resolved into two components of invariant magnitude:  $V_h$ , normal to the radius vector, and  $V_p$ , normal to the line of apsides. These components are important for both the classical and special hodograph. In the classical hodograph,  $V_h$  is the radius of the hodograph and  $V_p$  is the ordinate of its center. In the special hodograph,  $V_p$  is the radius of the hodograph and  $V_h$  is the abscissa of its center.

#### References

- 1 Cronin, J. L. and Schwartz, R. E., "Invariant two-body velocity components," *J. Aerospace Sci.* **29**, 1384-1385 (1962).
- 2 Altman, S. P. and Pistiner, J. S., "Hodograph transformations and mapping of the orbital conics," *ARS J.* **32**, 1109-1111 (1962).
- 3 Fang-Toh-Sun, "Some applications of the special hodograph for orbital motion," *Proceedings of the Third International Symposium on Rockets and Astronautics, Tokyo, 1962* (Yokendo, Bunkyo-Ku, Tokyo, 1962), pp. 381-396.
- 4 Altman, S. P. and Pistiner, J. S., "Polar hodograph for ballistic missile trajectories," *ARS J.* **31**, 1592-1594 (1961).

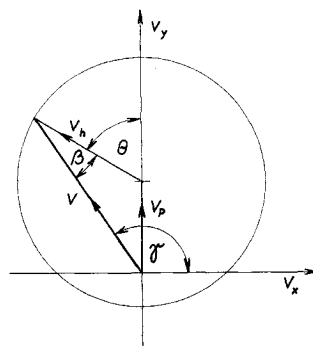
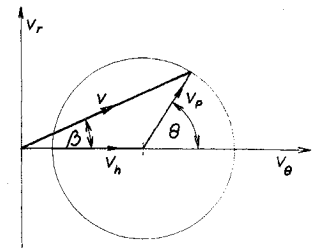


Fig. 2 Classical hodograph

Fig. 3 Special hodograph



<sup>5</sup> Altman, S. P. and Pistiner, J. S., "Hodograph analysis of the orbital transfer problem for coplanar nonaligned elliptical orbits," *ARS J.* **31**, 1217-1235 (1961).

<sup>6</sup> Fang-Toh-Sun, "A special hodograph for orbital motion," *Proceedings of the Second International Symposium on Rockets and Astronautics, Tokyo, 1960* (Yokendo, Bunkyo-Ku, Tokyo, 1961), pp. 163-189.

## Integration of the Velocity Equation of the Laminar Boundary Layer Including the Effects of Mass Transfer

H. L. EVANS\*

Imperial College, London, England

A METHOD is given for obtaining numerical solutions to the differential equation that governs fluid motion in a similar, laminar boundary layer. When some of the fluid flows in either direction through the wall, the problem contains two parameters, one being associated with the pressure gradient in the freestream while the other is a measure of the rate of mass transfer through the wall. The method of integration is quite general and gives high accuracy for any values of these parameters; it is particularly useful under conditions when other methods break down.

Spalding<sup>1</sup> has shown that the velocity equation of the uniform property, laminar boundary layer possesses similar solutions when the freestream velocity  $u_G$  obeys the relation

$$(du_G/dx) = C u_G^{2(\beta-1)/\beta} \quad (1)$$

where  $C$  and  $\beta$  are constants, the latter being the parameter associated with the freestream pressure gradient, and the  $x$  direction is parallel to the wall. When  $\eta$  the dimensionless distance from the wall, and  $f$  the dimensionless stream function are defined as

$$\eta = y \left( \frac{1}{\beta \nu} \frac{du_G}{dx} \right)^{1/2} \quad (2)$$

$$f = \frac{\psi}{u_G} \left( \frac{1}{\beta \nu} \frac{du_G}{dx} \right)^{1/2} \quad (3)$$

where  $y$  is the distance perpendicular to the wall,  $\nu$  is the kinematic viscosity, and  $\psi$  the ordinary stream function, the differential equation for similar boundary layers takes the form

$$f''' + ff'' + \beta(1 - f'^2) = 0 \quad (4)$$

with the boundary conditions

$$\begin{aligned} \eta = 0 \quad f = f_0 \quad f' = 0 \\ \eta \rightarrow \infty \quad f' \rightarrow 1 \end{aligned} \quad (5)$$

The primes in Eqs. (4) and (5) denote differentiation with

Received March 25, 1963.

\* Lecturer, Department of Mechanical Engineering.